

Neuroadaptive Fixed-Time Tracking Control of Full-State Constrained Strict-Feedback Nonlinear Systems with Actuator Faults

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Abstract—This paper investigates the problem of neuroadaptive fixed-time control for full-state constrained strict-feedback nonlinear systems subject to actuator faults. A fixed-time control strategy combined with barrier Lyapunov functions and neuroadaptive backstepping is proposed, and a neural network is employed to approximate the packaged unknown nonlinear terms and nonlinear actuator faults. By constructing barrier Lyapunov functions, it can be ensured that none of the strict-feedback systems' states will transgress their constraint bounds. Additionally, a fixed-time controller is designed such that all the signals in the closed-loop system are bounded, and the output is driven to track the reference signal to a small neighborhood within a fixed time. The benefits and feasibility of the proposed control method are also confirmed by simulations.

Index Terms—Fixed-time stability, Neuroadaptive, Strict-feedback systems, Actuator faults, Full-state constraints

I. INTRODUCTION

The problem of fixed-time control for strict feedback systems has attracted widespread considerable attention in the field of control theory. Actual system design often involves actuator failures, state constraints, and input saturation, among other factors, and the research results have mostly focused on stable system tracking under the abovementioned complex conditions[1], [2], [3], [4], [5], [6], [37].

In an actual engineering system, during long-term operation, actuator faults will inevitably occur. If a failure is not addressed in a timely and effective manner, the entire system may collapse. Therefore, it is necessary to establish a fault tolerant control (FTC) scheme. In view of this consideration, many neural network or fuzzy adaptive control schemes have been proposed, see, for example [7], [8], [9], [10], [34], [36] and the references cited therein. Nevertheless, the abovementioned system faults are linear faults, and these methods are not suitable for nonlinear faults. In addition, the status constraint, as a physical constraint, ensures the safety of equipment operation. To limit the system states to within the desired interval, a barrier Lyapunov function is introduced [11], [12], [13], [14], [15], [16], [38].

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Over the past decades, finite-time control has attracted interest from researchers due to its advantages, which include its fast convergence speed and strong anti-disturbance ability[17], [18], [19], [35]. The convergence time of finite time control depends on system's the initial state. In terms of degree, this phenomenon hinders the practical application of this method because an actual system's initial state cannot always be known in advance. Fortunately, reference [20] provides a fixed-time control method, where it is assumed that the convergence time is uniformly ultimately bounded and is independent of the initial state. In [21], aiming at the attitude tracking problem of four-rotor UAV (Unmanned Aerial Vehicle), a practical fixed-time disturbance rejection controller was proposed and considering the output constraint and input saturation of a pure-feedback system, a neuroadaptive fixed-time control scheme was provided in [22]. In [23], [24], [25], a fixed-time stable high-order nonlinear system was applied to the consensus for multi-agent systems.

To the best of our knowledge, the neuroadaptive fixed-time control of full-state constrained strict-feedback nonlinear systems with actuator faults cannot be designed or prespecified. The main contributions of this article in comparison with existing works are summarized as follows:

- A neuroadaptive controller that enables the nonlinear system to tracking a given desired trajectory within a fixed time and ensures that all variables in the closed-loop system are bounded is constructed.
- The neural network is used to approximate nonlinear actuator faults, which often appear during actual system operation.
- During the design process, a barrier Lyapunov function is introduced to constrain all state variables to within specified regions. Finally, the simulation results verify the effectiveness of the proposed control scheme.

II. PROBLEM FORMULATION AND PRELIMINARIES

A. System description

Consider the strict-feedback systems with actuator faults described by:

$$\begin{cases} \dot{x}_1 = x_2 + f_1(x_1) \\ \dot{x}_i = x_{i+1} + f_i(x_i) \\ \dot{x}_n = u + f_n(x_n) + \kappa(t - T_0)\chi(x, u) \\ y = x_1, i = 2, \dots, n - 1 \end{cases} \quad (1)$$

where $X = [x_1, \dots, x_n]$ represents the system state vector and is subject to $\|X\| < k_{b1}$ and $\|\dot{X}\| < k_{b2}$, where k_{b1} and k_{b2} are the given positive constants. $f_i(X)$ is an unknown smooth nonlinear function. u and y are control input and

output, respectively. $\kappa(t - T_0)$ represents the time function of the actuator failure at time T_0 , which is described as

$$\kappa(t - T_0) = \begin{cases} 1 - e^{-\nu(t-T_0)}, & t \geq T_0 \\ 0, & \text{otherwise} \end{cases} \quad (2)$$

where $\nu > 0$ is the evolution rate of the unknown fault. $\chi(q, u)$ is an unknown fault function.

Remark 1: This paper is concerned with nonlinear actuator faults, which are different from the linear faults discussed in references [7], [26], [27] and are more in line with actual actuator faults.

Control objective: to design an adaptive NN (Neural Network) control law for the strict feedback system (1) under actuator failure and full state constraints to ensure that the output x_1 of the strict feedback system (1) is along the desired trajectory x_d while the tracking error is stabilized in a small residual set at a fixed time.

B. Preliminaries

To convenient for the control system design, some assumptions and lemmas are imposed on system (1).

Assumption 1 ([28]): The reference signal x_d and its derivatives \dot{x}_d and \ddot{x}_d are bounded.

Assumption 2 ([29]): There is an unknown non-negative function $g(x, u)$, and

$$|f_n(x_n) + \kappa(t - T_0)\chi(x, u)| \leq g(x, u) \quad (3)$$

Lemma 1: Radial basis function neural networks (RBFNN) can approximate unknown continuous nonlinear functions $f(Z)$ with arbitrary precision on a compact set Ω .

$$f(Z) = W^T \Phi(Z) + \delta(Z) \quad (4)$$

where $W = [W_1, W_2, \dots, W_\ell]$ is the optimal weight of RBFNN, $\delta(Z)$ is the approximation error, satisfies $|\delta(Z)| \leq \varepsilon$, and $\varepsilon > 0$ is a constant. ℓ is the number of NN nodes. $\Phi(Z) = [\Phi_1(Z), \Phi_2(Z), \dots, \Phi_\ell(Z)]$ is a known and bounded basis function, and chosen as the Gaussian function form

$$\Phi_i(Z) = \exp\left[-\frac{(Z - \varsigma_i)^T(Z - \varsigma_i)}{r_i^2}\right] \quad (5)$$

where $\varsigma_i = [\varsigma_{i1}, \varsigma_{i2}, \dots, \varsigma_{i\ell}]^T$ denotes the center of the receptive field, and r_i represents the width of the Gaussian function.

Remark 2: Here we define an unknown constant Θ_i as $\Theta_i = \|W_i\|^2/b, i = 1, 2, \dots, n$.

Definition 1 ([30]): Consider the nonlinear dynamical system

$$\dot{x} = f(x), x(0) = x_0 \quad (6)$$

where x is the system state and $f(x)$ is a smooth nonlinear function. Then, assume that the origin is an equilibrium point.

Lemma 2([22]): For system (1), suppose there is a Lyapunov function $V(x)$, so that the following inequality holds

$$\dot{V}(x) \leq -\rho_1 V^\alpha(x) - \rho_2 V^\beta(x) + \zeta \quad (7)$$

where $\rho_1 > 0, \rho_2 > 0, \zeta > 0, \alpha \in (0, 1)$, and $\beta \in (1, \infty)$. Then the origin of system (1) exhibits practical fixed-time stability and the settling time satisfies

$$T \leq T_{max} := \frac{1}{\rho_1\theta(1-\alpha)} + \frac{1}{\rho_2\theta(\beta-1)} \quad (8)$$

where θ is a constant and satisfies $\theta \in (0, 1)$. The residual set of the solution in (7) is expressed as

$$x \in \{V(x) \leq \min\left\{\left(\frac{\zeta}{(1-\theta)\rho_2}\right)^{\frac{1}{\beta}}, \left(\frac{\zeta}{(1-\theta)\rho_1}\right)^{\frac{1}{\alpha}}\right\}\} \quad (9)$$

Remark 3: From (8), the advantage of fixed-time control over finite-time control is that the upper bound of the fixed-time convergence time is independent of the initial conditions and is only related to the design parameters.

Lemma 3([31]): For any scalars $\Gamma_i \in R, i = 1, 2, \dots, N, 0 < \rho_1 < 1$, and $\rho_2 > 1$. There holds

$$\left(\sum_{i=1}^N |\Gamma_i|\right)^{\rho_1} \leq \sum_{i=1}^N |\Gamma_i|^{\rho_1} \quad (10)$$

$$\left(\sum_{i=1}^N |\Gamma_i|\right)^{\rho_2} \leq N^{\rho_2-1} \sum_{i=1}^N |\Gamma_i|^{\rho_2} \quad (11)$$

Lemma 4([32]): For any positive real number a, b , and $\psi(\mu, \nu) > 0$, the following relationship holds

$$|\mu|^a |\nu|^b \leq \frac{a\psi |\mu|^{a+b}}{a+b} + \frac{b\psi^{-\frac{a}{b}} |\nu|^{a+b}}{a+b} \quad (12)$$

III. CONTROL SCHEME

To realize the stability analysis of the system (1), introduce the change of coordinates:

$$\begin{aligned} z_1 &= x_1 - x_d \\ z_i &= x_i - s_i \\ \varpi_i &= s_i - \varrho_{i-1}, i = 2, \dots, n \end{aligned} \quad (13)$$

where z_1 and z_i are virtual error surfaces, ϱ_{i-1} is virtual control signals, and ϖ_i is error signals. s_i denotes the state variables, which are obtained by first-order filtering of the virtual control signal ϱ_{i-1} . Therefore, the stability of ϖ_i is constructed and analyzed using n steps based on the backstepping technique.

Step 1: It follows from (1) and (13), that

$$\begin{aligned} \dot{z}_1 &= \dot{x}_1 - \dot{x}_d \\ &= x_2 + f_1(x_1) - \dot{x}_d \end{aligned} \quad (14)$$

Consider the following Lyapunov function candidate as

$$V_1 = \frac{1}{2} \log \frac{k_{b1}^2}{k_{b1}^2 - z_1^2} + \frac{b}{2\gamma_1} \tilde{\Theta}_1^2 \quad (15)$$

where γ_1 is the design parameter and $\tilde{\Theta}_1 = \Theta_1 - \hat{\Theta}_1, \hat{\Theta}_1$ is the estimated value of Θ_1 .

Taking the derivative of V_1 with respect to (w.r.t.) z_1 and $\tilde{\Theta}_1$, one has

$$\begin{aligned} \dot{V}_1 &= \frac{z_1}{k_{b1}^2 - z_1^2} (x_2 + f_1(x_1) - \dot{x}_d) - \frac{b}{\gamma_1} \tilde{\Theta}_1 \dot{\hat{\Theta}}_1 \\ &= \frac{z_1}{k_{b1}^2 - z_1^2} (z_2 + \varpi_2 + \varrho_1 + \hat{f}_1(Z_1)) - \frac{1}{2} \left(\frac{z_1}{k_{b1}^2 - z_1^2}\right)^2 \\ &\quad - \frac{b}{\gamma_1} \tilde{\Theta}_1 \dot{\hat{\Theta}}_1 \end{aligned} \quad (16)$$

where $\hat{f}_1(Z_1) = f_1(x_1) - \dot{x}_d + \frac{1}{2} \left(\frac{z_1}{k_{b1}^2 - z_1^2}\right)$. According to (4), one has

$$\hat{f}_1(Z_1) = W_1^T \Phi_1(Z_1) + \delta_1 \quad (17)$$

where $\delta_1 \leq \varepsilon_1$. By utilizing Young's inequality, one has

$$\begin{aligned} \hat{f}_1(Z_1) \frac{z_1}{k_{b1}^2 - z_1^2} &\leq \frac{b\Theta_1}{2a_1^2} \left(\frac{z_1}{k_{b1}^2 - z_1^2} \right)^2 \Phi_1^T(Z_1) \Phi_1(Z_1) + \frac{a_1^2}{2} \\ &\quad + bc_{13} \left(\frac{z_1}{k_{b1}^2 - z_1^2} \right)^2 + \frac{\varepsilon_1^2}{4bc_{13}} \end{aligned} \quad (18)$$

$$\frac{z_1}{k_{b1}^2 - z_1^2} \varpi_2 \leq \frac{1}{2} \left(\frac{z_1}{k_{b1}^2 - z_1^2} \right)^2 + \frac{1}{2} \varpi_2^2 \quad (19)$$

By combining (17), (18), and (19), it is readily shown that

$$\begin{aligned} \dot{V}_1 &\leq \frac{b\Theta_1}{2a_1^2} \left(\frac{z_1}{k_{b1}^2 - z_1^2} \right)^2 \Phi_1^T(Z_1) \Phi_1(Z_1) \\ &\quad + \frac{a_1^2}{2} + bc_{13} \left(\frac{z_1}{k_{b1}^2 - z_1^2} \right)^2 + \frac{\varepsilon_1^2}{4bc_{13}} + \frac{1}{2} \varpi_2^2 \\ &\quad + \frac{z_1}{k_{b1}^2 - z_1^2} \varrho_1 - \frac{b}{\gamma_1} \dot{\Theta}_1 \dot{\Theta}_1 + \frac{z_1}{k_{b1}^2 - z_1^2} z_2 \end{aligned} \quad (20)$$

To proceed, we define the virtual control input ϱ_1 as:

$$\begin{aligned} \varrho_1 &= -c_{11} \left(\frac{z_1^2}{k_{b1}^2 - z_1^2} \right)^{\frac{\alpha-1}{2}} z_1 - c_{12} \left(\frac{z_1^2}{k_{b1}^2 - z_1^2} \right)^{\beta-1} z_1 \\ &\quad - \frac{\dot{\Theta}_1}{2a_1^2} \left(\frac{z_1}{k_{b1}^2 - z_1^2} \right)^2 \Phi_1^T(Z_1) \Phi_1(Z_1) - c_{13} \left(\frac{z_1}{k_{b1}^2 - z_1^2} \right) \end{aligned} \quad (21)$$

With (21), (20) can be continued as follows:

$$\begin{aligned} \dot{V}_1 &\leq \frac{b\dot{\Theta}_1}{\gamma_1} \left[\frac{\gamma_1}{2a_1^2} \left(\frac{z_1}{k_{b1}^2 - z_1^2} \right)^2 \Phi_1^T(Z_1) \Phi_1(Z_1) - \dot{\Theta}_1 \right] \\ &\quad + \frac{a_1^2}{2} + \frac{\varepsilon_1^2}{4bc_{13}} + \frac{1}{2} \varpi_2^2 - c_{11} \left(\frac{z_1^2}{k_{b1}^2 - z_1^2} \right)^{\frac{\alpha+1}{2}} \\ &\quad - c_{12} \left(\frac{z_1^2}{k_{b1}^2 - z_1^2} \right)^\beta + \frac{z_1}{k_{b1}^2 - z_1^2} z_2 \end{aligned} \quad (22)$$

The parameter adaptive law of $\dot{\Theta}_1$ is designed as

$$\dot{\Theta}_1 = \frac{\gamma_1}{2a_1^2} \left(\frac{z_1}{k_{b1}^2 - z_1^2} \right)^2 \Phi_1^T(Z_1) \Phi_1(Z_1) - 2r_1 \dot{\Theta}_1 \quad (23)$$

Substituting the virtual control input and adaptive law into (22), it is readily seen that

$$\begin{aligned} \dot{V}_1 &\leq \frac{2br_1 \dot{\Theta}_1 \Theta_1}{\gamma_1} + \frac{a_1^2}{2} + \frac{\varepsilon_1^2}{4bc_{13}} + \frac{1}{2} \varpi_2^2 \\ &\quad - c_{11} \left(\frac{z_1^2}{k_{b1}^2 - z_1^2} \right)^{\frac{\alpha+1}{2}} - c_{12} \left(\frac{z_1^2}{k_{b1}^2 - z_1^2} \right)^\beta + \frac{z_1}{k_{b1}^2 - z_1^2} z_2 \end{aligned} \quad (24)$$

From Lemma 3, it hold that

$$\begin{aligned} \frac{2br_1 \dot{\Theta}_1 \Theta_1}{\gamma_1} &\leq -\frac{br_1}{\gamma_1} \dot{\Theta}_1^2 + \frac{br_1}{\gamma_1} \Theta_1^2 \\ &= -\frac{br_1}{2\gamma_1} \dot{\Theta}_1^2 - \frac{br_1}{2\gamma_1} \dot{\Theta}_1^2 + \frac{br_1}{\gamma_1} \Theta_1^2 \end{aligned} \quad (25)$$

then, according to Lemma 4, ones has

$$-\frac{br_1}{2\gamma_1} \dot{\Theta}_1^2 \leq -r_1 \left(\frac{b}{2\gamma_1} \dot{\Theta}_1^2 \right)^{\frac{1+\alpha}{2}} + r_1 \left(1 - \frac{1+\alpha}{2} \right) \psi_1 \quad (26)$$

and

$$-\frac{br_1}{2\gamma_1} \dot{\Theta}_1^2 \leq -r_1 \left(\frac{b}{2\gamma_1} \dot{\Theta}_1^2 \right)^\beta + r_1 (1 - \beta) \psi_2 \quad (27)$$

Invoking (26) and (27), (24) can be rewritten as

$$\begin{aligned} \dot{V}_1 &\leq -c_{11} \left(\frac{z_1^2}{k_{b1}^2 - z_1^2} \right)^{\frac{\alpha+1}{2}} - c_{12} \left(\frac{z_1^2}{k_{b1}^2 - z_1^2} \right)^\beta \\ &\quad - r_1 \left(\frac{b}{2\gamma_1} \dot{\Theta}_1^2 \right)^{\frac{1+\alpha}{2}} - r_1 \left(\frac{b}{2\gamma_1} \dot{\Theta}_1^2 \right)^\beta + \frac{a_1^2}{2} + \frac{\varepsilon_1^2}{4bc_{13}} \\ &\quad + \frac{1}{2} \varpi_2^2 + \frac{z_1}{k_{b1}^2 - z_1^2} z_2 + r_1 b \left(1 - \frac{1+\alpha}{2} \right) \psi_1 \\ &\quad + r_1 b (1 - \beta) \psi_2 + \frac{br_1}{\gamma_1} \Theta_1^2 \end{aligned} \quad (28)$$

Then, (30) can be rewritten as

$$\dot{V}_1 \leq -\lambda_{11} V_1^{\frac{1+\alpha}{2}} - \lambda_{12} 2^{1-\beta} V_1^\beta + \frac{z_1}{k_{b1}^2 - z_1^2} z_2 + \Delta_1 \quad (29)$$

where $\lambda_{11} = \min\{c_{11} 2^{\frac{1+\alpha}{2}}, r_1\}$, $\lambda_{12} = \min\{c_{12} 2^\beta, r_1\}$ and $\Delta_1 = \frac{a_1^2}{2} + \frac{\varepsilon_1^2}{4bc_{13}} + \frac{1}{2} \varpi_2^2 + \frac{br_1}{\gamma_1} \Theta_1^2 + r_1 b \left(1 - \frac{1+\alpha}{2} \right) \psi_1 + r_1 b (1 - \beta) \psi_2$.

To avoid the problem of "explosion of complexity", we introduce a new state variable s_2 and letting ϱ_1 pass through a first-order filter with a time constant τ_2 that yields

$$\tau_2 \dot{s}_2 + s_2 = \varrho_1, s_2(0) = \varrho_1(0) \quad (30)$$

where $\tau_2 > 0$ is a constant. Combined with (14), we can get

$$\dot{s}_2 = \frac{\varrho_1 - s_2}{\tau_2} = -\frac{\varpi_2}{\tau_2} \quad (31)$$

then

$$\dot{\varpi}_2 = -\frac{\varpi_2}{\tau_2} + N_2(\cdot) \quad (32)$$

where $N_2(\cdot) = \dot{\varrho}_1$, specifically expressed as

$$\begin{aligned} N_2(\cdot) &= -\rho_{11} (2\alpha - 1) \left(\frac{1}{2} \right)^\alpha z_1^{2\alpha-2} \dot{z}_1 \\ &\quad - \rho_{21} (2\beta - 1) \left(\frac{1}{2} \right)^\beta z_1^{2\beta-2} \dot{z}_1 - \dot{z}_1 \\ &\quad - \dot{W}_1^T \Phi_1(x_1) + \hat{W}_1^T \frac{\partial \Phi_1(x_1)}{\partial x_1} \dot{x}_1 + \ddot{x}_d \end{aligned} \quad (33)$$

Step $i(i=2, \dots, n-1)$: According to (1), (4), and (14), the time derivative of z_i can be obtained by

$$\begin{aligned} \dot{z}_i &= x_{i+1} + f_i(x_i) - \dot{s}_i \\ &= z_{i+1} + \varpi_{i+1} + \varrho_i + f_i(x_i) - \dot{s}_i \end{aligned} \quad (34)$$

Consider the Lyapunov function candidate V_i as

$$V_i = V_{i-1} + \frac{1}{2} \log \frac{k_{bi}^2}{k_{bi}^2 - z_i^2} + \frac{1}{2} \varpi_i^2 + \frac{b}{2\gamma_i} \tilde{\Theta}_i^2 \quad (35)$$

The time derivative of V_i along (30) can be derived as

$$\begin{aligned} \dot{V}_i &= \dot{V}_{i-1} + \frac{z_i}{k_{bi}^2 - z_i^2} (f_i(x_i) + z_{i+1} + \varpi_{i+1} \\ &\quad + \varrho_i - \dot{s}_i) + \varpi_i \dot{\varpi}_i - \frac{b}{\gamma_i} \dot{\Theta}_i \dot{\Theta}_i \end{aligned} \quad (36)$$

According to Lemma 1, one has

$$\hat{f}_i(Z_i) = W_i^T \Phi_i(Z_i) + \delta_i \quad (37)$$

where $\hat{f}_i(Z_i) = f_i(x_i) - \dot{s}_i + \frac{1}{2} \left(\frac{z_i}{k_{bi}^2 - z_i^2} \right)$. By using Young's inequality, we can get

$$\begin{aligned} \hat{f}_i(Z_i) \frac{z_i}{k_{bi}^2 - z_i^2} &\leq \frac{b\Theta_i}{2a_i^2} \left(\frac{z_i}{k_{bi}^2 - z_i^2} \right)^2 \Phi_i^T(Z_i) \Phi_i(Z_i) \\ &\quad + \frac{a_i^2}{2} + bc_{i3} \left(\frac{z_i}{k_{bi}^2 - z_i^2} \right)^2 + \frac{\varepsilon_i^2}{4bc_{i3}} \end{aligned} \quad (38)$$

$$\frac{z_i}{k_{bi}^2 - z_i^2} \varpi_{i+1} \leq \frac{1}{2} \left(\frac{z_i}{k_{bi}^2 - z_i^2} \right)^2 + \frac{1}{2} \varpi_{i+1}^2 \quad (39)$$

From (37), (38), and (39), it is derived that

$$\begin{aligned} \dot{V}_i &\leq \dot{V}_{i-1} + \frac{b\Theta_i}{2a_i^2} \left(\frac{z_i}{k_{bi}^2 - z_i^2} \right)^2 \Phi_i^T(Z_i) \Phi_i(Z_i) + \frac{a_i^2}{2} \\ &+ bc_{i3} \left(\frac{z_i}{k_{bi}^2 - z_i^2} \right)^2 + \frac{\varepsilon_i^2}{4bc_{i3}} + \frac{1}{2} \varpi_{i+1}^2 + \frac{z_i}{k_{bi}^2 - z_i^2} z_{i+1} \\ &+ \frac{z_i}{k_{bi}^2 - z_i^2} \varrho_i + \varpi_i \left(-\frac{\varpi_i}{\tau_i} + N_i(\cdot) \right) - \frac{b}{\gamma_i} \tilde{\Theta}_i \dot{\tilde{\Theta}}_i \end{aligned} \quad (40)$$

By using Young's inequality, we have

$$\varpi_i N_i(\cdot) \leq \frac{1}{2\phi} \varpi_i^2 N_i^2 + \frac{1}{2} \phi \quad (41)$$

where ϕ is a positive constant. The virtual control input ϱ_i and adaptive law for $\hat{\Theta}_i$ are designed as

$$\varrho_i = -c_{i1} \left(\frac{z_i^2}{k_{bi}^2 - z_i^2} \right)^{\frac{\alpha-1}{2}} z_i - c_{i2} \left(\frac{z_i^2}{k_{bi}^2 - z_i^2} \right)^{\beta-1} z_i \quad (42)$$

$$- \frac{\hat{\Theta}_i}{2a_i^2} \left(\frac{z_i}{k_{bi}^2 - z_i^2} \right)^2 \Phi_i^T(Z_i) \Phi_i(Z_i) - c_{i3} \left(\frac{z_i}{k_{bi}^2 - z_i^2} \right)$$

$$\dot{\hat{\Theta}}_i = \frac{\gamma_i}{2a_i^2} \left(\frac{z_i}{k_{bi}^2 - z_i^2} \right)^2 \Phi_i^T(Z_i) \Phi_i(Z_i) - 2r_i \hat{\Theta}_i \quad (43)$$

Then, (40) can be rewritten as

$$\begin{aligned} \dot{V}_i &\leq \sum_{j=1}^j \frac{2br_j \tilde{\Theta}_j \hat{\Theta}_j}{\gamma_j} + \sum_{j=1}^i \frac{a_j^2}{2} + \sum_{j=1}^i \frac{\varepsilon_j^2}{4bc_{j3}} \\ &+ \frac{z_i}{k_{bi}^2 - z_i^2} z_{i+1} - \sum_{j=1}^i c_{j1} \left(\frac{z_j^2}{k_{bj}^2 - z_j^2} \right)^{\frac{\alpha+1}{2}} \\ &- \sum_{j=1}^i c_{j2} \left(\frac{z_j^2}{k_{bj}^2 - z_j^2} \right)^{\beta} + \frac{1}{2} \phi + \sum_{j=1}^i \varpi_{j+1}^2 \\ &- \sum_{j=2}^i \left(\frac{1}{\tau_i} - \frac{1}{2\phi} N_i^2 - 1 \right) \varpi_j^2 \end{aligned} \quad (44)$$

According to Lemma 3, we have

$$\begin{aligned} \frac{2br_i \tilde{\Theta}_i \hat{\Theta}_i}{\gamma_i} &\leq -\frac{br_i}{\gamma_i} \tilde{\Theta}_i^2 + \frac{br_i}{\gamma_i} \Theta_i^2 \\ &= -\frac{br_i}{2\gamma_i} \tilde{\Theta}_i^2 - \frac{br_i}{2\gamma_i} \tilde{\Theta}_i^2 + \frac{br_i}{\gamma_i} \Theta_i^2 \end{aligned} \quad (45)$$

Then, Upon using Lemma 4, it is shown that

$$-\frac{br_i}{2\gamma_i} \tilde{\Theta}_i^2 \leq -r_i \left(\frac{b}{2\gamma_i} \tilde{\Theta}_i^2 \right)^{\frac{1+\alpha}{2}} + r_i \left(1 - \frac{1+\alpha}{2} \right) \psi_{i1} \quad (46)$$

and

$$-\frac{br_i}{2\gamma_i} \tilde{\Theta}_i^2 \leq -r_i \left(\frac{b}{2\gamma_i} \tilde{\Theta}_i^2 \right)^{\beta} + r_i (1 - \beta) \psi_{i2} \quad (47)$$

$$\begin{aligned} -\left(\frac{2}{\tau_i} - \frac{1}{\phi} N_i^2 - 2 \right) \frac{1}{2} \varpi_j^2 &\leq -\left(\frac{2}{\tau_i} - \frac{1}{\phi} N_i^2 - 2 \right) \left(\frac{1}{2} \varpi_j^2 \right)^{\frac{1+\alpha}{2}} \\ &+ \left(\frac{2}{\tau_i} - \frac{1}{\phi} N_i^2 - 2 \right) \left(1 - \frac{1+\alpha}{2} \right) \psi_{i3} \end{aligned} \quad (48)$$

$$\begin{aligned} -\left(\frac{2}{\tau_i} - \frac{1}{\phi} N_i^2 - 2 \right) \frac{1}{2} \varpi_j^2 &\leq -\left(\frac{2}{\tau_i} - \frac{1}{\phi} N_i^2 - 2 \right) \left(\frac{1}{2} \varpi_j^2 \right)^{\beta} \\ &+ \left(\frac{2}{\tau_i} - \frac{1}{\phi} N_i^2 - 2 \right) (1 - \beta) \psi_{i4} \end{aligned} \quad (49)$$

The (44) can be rewritten as

$$\dot{V}_i \leq -\lambda_{i1} V_i^{\frac{1+\alpha}{2}} - \lambda_{i2} 2^{1-\beta} V_i^{\beta} + \frac{z_i}{k_{bi}^2 - z_i^2} z_{i+1} + \Delta_i \quad (50)$$

where $\Delta_i = \sum_{j=1}^i \frac{a_j^2}{2} + \sum_{j=1}^i \frac{\varepsilon_j^2}{4bc_{j3}} + \sum_{j=1}^i \varpi_{j+1}^2 + \sum_{j=1}^i \frac{br_i}{\gamma_i} \Theta_i^2 + \sum_{j=1}^i r_i b \left(1 - \frac{1+\alpha}{2} \right) \psi_i + \sum_{j=1}^i r_i b (1 - \beta) \psi_{i+1}$, $\lambda_{i1} = \min\{\sum_{j=1}^i c_{i1} 2^{\frac{1+\alpha}{2}}\}$, $\sum_{j=1}^i r_i$, $\lambda_{i2} = \min\{\sum_{j=1}^i c_{i2} 2^{\beta}, \sum_{j=1}^i r_i\}$.

To avoid repeatedly differentiating ϱ_i , we define the first-order filter as

$$\tau_{i+1} \dot{s}_{i+1} + s_{i+1} = \varrho_i, s_{i+1}(0) = \varrho_i(0) \quad (51)$$

where $\tau_{i+1} > 0$ is a constant. Combined with (13), we can get

$$\dot{s}_{i+1} = \frac{\varrho_i - s_{i+1}}{\tau_{i+1}} = -\frac{\varpi_{i+1}}{\tau_{i+1}} \quad (52)$$

then

$$\dot{\varpi}_{i+1} = -\frac{\varpi_{i+1}}{\tau_{i+1}} + N_{i+1}(\cdot) \quad (53)$$

where N_{i+1} is a continuous function, and specifically expressed as

$$\begin{aligned} N_{i+1}(\cdot) &= -\rho_{11} (2\alpha - 1) \left(\frac{1}{2} \right)^{\alpha} z_1^{2\alpha-2} \dot{z}_1 \\ &- \rho_{21} (2\beta - 1) \left(\frac{1}{2} \right)^{\beta} z_1^{2\beta-2} \dot{z}_1 - \dot{z}_1 \end{aligned} \quad (54)$$

$$- \hat{W}_1^T \Phi_1(x_1) + \hat{W}_1^T \frac{\partial \Phi_1(x_1)}{\partial x_1} \dot{x}_1 + \ddot{y}_d$$

Step n : According to (1) and (13), we can get

$$\dot{z}_n = u + f_n(x_n) + \kappa(t - T_0) \chi(x, u) - \dot{s}_n \quad (55)$$

Letting $G(x, u) = g(x, u) - \dot{s}_n + \frac{1}{2} \left(\frac{z_n}{k_{bn}^2 - z_n^2} \right)$. Then, from Assumption 2 and Lemma 1, ones has

$$G(x, u) = W_n^T \Phi_n(Z_n) + \delta_n \quad (56)$$

Choose the Lyapunov function candidate as

$$V_n = V_{n-1} + \frac{1}{2} \log \frac{k_{bn}^2}{k_{bn}^2 - z_n^2} + \frac{1}{2} \varpi_n^2 + \frac{b}{2\gamma_n} \tilde{\Theta}_n^2 \quad (57)$$

The time derivative of V_n along (55) and (56) can be derived as

$$\begin{aligned} \dot{V}_n &= \dot{V}_{n-1} + \frac{z_n}{k_{bn}^2 - z_n^2} (W_n^T \Phi_n(Z_n) + \delta_n \\ &+ u) + \varpi_n \dot{\varpi}_n - \frac{b}{\gamma_n} \tilde{\Theta}_n \dot{\tilde{\Theta}}_n \end{aligned} \quad (58)$$

By using Young's inequality, we have

$$\begin{aligned} G(Z_n) \frac{z_n}{k_{bn}^2 - z_n^2} &\leq \frac{b\Theta_n}{2a_n^2} \left(\frac{z_n}{k_{bn}^2 - z_n^2} \right)^2 \Phi_n^T(Z_n) \Phi_n(Z_n) \\ &+ \frac{a_n^2}{2} + bc_{n3} \left(\frac{z_n}{k_{bn}^2 - z_n^2} \right)^2 + \frac{\varepsilon_n^2}{4bc_{n3}} \end{aligned} \quad (59)$$

Hence, (58) becomes

$$\begin{aligned} \dot{V}_n &\leq \dot{V}_{n-1} + \frac{b\Theta_n}{2a_n^2} \left(\frac{z_n}{k_{bn}^2 - z_n^2} \right)^2 \Phi_n^T(Z_n) \Phi_n(Z_n) + \frac{a_n^2}{2} \\ &+ bc_{n3} \left(\frac{z_n}{k_{bn}^2 - z_n^2} \right)^2 + \frac{\varepsilon_n^2}{4bc_{n3}} \\ &+ \frac{z_n}{k_{bn}^2 - z_n^2} u + \varpi_n \left(-\frac{\varpi_n}{\tau_i} + N_n(\cdot) \right) - \frac{b}{\gamma_n} \tilde{\Theta}_n \dot{\tilde{\Theta}}_n \end{aligned} \quad (60)$$

By using Young's inequality, we have

$$\varpi_n N_n(\cdot) \leq \frac{1}{2\phi} \varpi_n^2 N_n^2 + \frac{1}{2}\phi \quad (61)$$

We design the control input u and the parameter adaptive law of $\hat{\Theta}_n$ as

$$u = -c_{i1} \left(\frac{z_n^2}{k_{bn}^2 - z_n^2} \right)^{\frac{\alpha-1}{2}} z_n - c_{n2} \left(\frac{z_n^2}{k_{bn}^2 - z_n^2} \right)^{\beta-1} z_n - \frac{\hat{\Theta}_n}{2a_n^2} \frac{z_n}{k_{bn}^2 - z_n^2} \Phi_n^T(Z_n) \Phi_n(Z_n) - c_{n3} \left(\frac{z_n}{k_{bn}^2 - z_n^2} \right) - z_{n-1} \quad (62)$$

$$\dot{\hat{\Theta}}_n = \frac{\gamma_n}{2a_n^2} \left(\frac{z_n}{k_{bn}^2 - z_n^2} \right)^2 \Phi_n^T(Z_n) \Phi_n(Z_n) - 2r_n \hat{\Theta}_n \quad (63)$$

Applying (62) and (63), we obtain

$$\begin{aligned} \dot{V}_n &\leq \frac{2br_n \tilde{\Theta}_n \hat{\Theta}_n}{\gamma_n} + \sum_{i=1}^n \frac{a_i^2}{2} + \sum_{i=1}^n \frac{\varepsilon_i^2}{4bc_{i3}} \\ &\quad - \sum_{i=1}^n c_{i1} \left(\frac{z_i^2}{k_{bi}^2 - z_i^2} \right)^{\frac{\alpha+1}{2}} - \sum_{i=1}^n c_{i2} \left(\frac{z_i^2}{k_{bi}^2 - z_i^2} \right)^\beta \\ &\quad + \frac{1}{2}\phi - \sum_{i=2}^n \left(\frac{1}{\tau_i} - \frac{1}{2\phi} N_i^2 - 1 \right) \varpi_i^2 \end{aligned} \quad (64)$$

From Lemma 3, we have

$$\begin{aligned} \frac{2br_n \tilde{\Theta}_n \hat{\Theta}_n}{\gamma_n} &\leq -\frac{br_n}{\gamma_n} \tilde{\Theta}_n^2 + \frac{br_n}{\gamma_n} \Theta_n^2 \\ &= -\frac{br_n}{2\gamma_n} \tilde{\Theta}_n^2 - \frac{br_n}{2\gamma_n} \tilde{\Theta}_n^2 + \frac{br_n}{\gamma_n} \Theta_n^2 \end{aligned} \quad (65)$$

Then, according to Lemma 4, ones has

$$-\frac{br_n}{2\gamma_n} \tilde{\Theta}_n^2 \leq -r_n \left(\frac{b}{2\gamma_n} \tilde{\Theta}_n^2 \right)^{\frac{1+\alpha}{2}} + r_n \left(1 - \frac{1+\alpha}{2} \right) \psi_{n1} \quad (66)$$

and

$$-\frac{br_n}{2\gamma_n} \tilde{\Theta}_n^2 \leq -r_n \left(\frac{b}{2\gamma_n} \tilde{\Theta}_n^2 \right)^\beta + r_n (1 - \beta) \psi_{n2} \quad (67)$$

Same as (66) and (67), we have

$$\begin{aligned} -\left(\frac{2}{\tau_n} - \frac{1}{\phi} N_n^2 - 2 \right) \frac{1}{2} \varpi_n^2 &\leq -\left(\frac{2}{\tau_n} - \frac{1}{\phi} N_n^2 - 2 \right) \left(\frac{1}{2} \varpi_n^2 \right)^{\frac{1+\alpha}{2}} \\ &\quad + \left(\frac{2}{\tau_n} - \frac{1}{\phi} N_n^2 - 2 \right) \left(1 - \frac{1+\alpha}{2} \right) \psi_{n3} \end{aligned} \quad (68)$$

$$\begin{aligned} -\left(\frac{2}{\tau_n} - \frac{1}{\phi} N_n^2 - 2 \right) \frac{1}{2} \varpi_n^2 &\leq -\left(\frac{2}{\tau_n} - \frac{1}{\phi} N_n^2 - 2 \right) \left(\frac{1}{2} \varpi_n^2 \right)^\beta \\ &\quad + \left(\frac{2}{\tau_n} - \frac{1}{\phi} N_n^2 - 2 \right) (1 - \beta) \psi_{n4} \end{aligned} \quad (69)$$

Consequently, (64) becomes

$$\dot{V}_n \leq -\lambda_{n1} V_n^{\frac{1+\alpha}{2}} - \lambda_{n2} 2^{1-\beta} V_n^\beta + \Delta \quad (70)$$

where $\lambda_{n1} = \min\{c_{i1} 2^{\frac{1+\alpha}{2}}, \dots, c_{n1} 2^{\frac{1+\alpha}{2}}, r_1, \dots, r_n, \frac{2}{\tau_2} - \frac{1}{\phi} N_2^2 - 2, \dots, \frac{2}{\tau_n} - \frac{1}{\phi} N_n^2 - 2\}$, $\lambda_{i2} = \min\{c_{i2} 2^\beta, \dots, c_{n2} 2^\beta, r_1, \dots, r_n, \frac{2}{\tau_2} - \frac{1}{\phi} N_2^2 - 2, \dots, \frac{2}{\tau_n} - \frac{1}{\phi} N_n^2 - 2\}$ and $\Delta = \sum_{i=1}^n \frac{a_i^2}{2} + \sum_{i=1}^n \frac{\varepsilon_i^2}{4bc_{i3}} + \frac{1}{2}\varpi_2^2 + \frac{br_1}{\gamma_1} \Theta_1^2 + r_1 b \left(1 - \frac{1+\alpha}{2} \right) \psi_1 + r_1 b (1 - \beta) \psi_2$.

Using Lemma 2, it is guaranteed that all signals in the system (1) are practically fixed-time stable and that the signals converge within the set tight set.

$$x \in \{V(x) \leq \min\left\{ \left(\frac{2^{\beta-1} \Delta}{\lambda_{n1} 2^{1-\theta}} \right)^{\frac{1}{\beta}}, \left(\frac{\Delta}{\lambda_{n1} (1-\theta)} \right)^{\frac{2}{1+\alpha}} \right\}\} \quad (71)$$

The tracking error converges to a small neighborhood near the origin, satisfying

$$|y - x_d| \leq k_{b1} \left[1 - e^{-2 \left(\frac{\Delta}{(1-\theta)\lambda_{n1}} \right)^{\frac{2}{1+\alpha}}} \right]^{\frac{1}{2}} \quad (72)$$

and the fixed time is selected with

$$T \leq T_{\max} := \frac{2}{\lambda_{n1} \theta (1 - \alpha)} + \frac{2^{\beta-1}}{\lambda_{n2} \theta (\beta - 1)} \quad (73)$$

Based on the above discussions, we are going to express the main results of this paper with the following theorem.

Theorem 1: For the nonlinear system (1) with virtual control inputs (21) and (42), the actual control input is described in (62) and the adaptive laws are discussed in (23), (43), and (63). Under Assumptions 1-2, i) the closed loop system has semi-globally practical fixed-time stability, and ii) the tracking error $|y - x_d|$ converges to a small residual set in a fixed time.

Remark 4: To avoid divergence of the control input u , the range of values of α is narrowed to $\alpha \in (0.6, 1)$.

IV. SIMULATION VERIFICATION

A. Mathematical example

Consider the following strict-feedback nonlinear systems:

$$\begin{cases} \dot{x}_1 = x_2 + 0.1x_1^2 \\ \dot{x}_i = 0.1x_1 x_2 - 0.2x_1 + u + \kappa(t - T_0) \chi(x, u) \\ y = x_1 \end{cases} \quad (74)$$

The initial state of the system is $x_1(0) = 0.2$ and $x_2(0) = -0.3$. The desired trajectory of the system tracking target is $x_d = 0.5 \sin(t)$. The states are constrained by $k_{b1} = 0.6$ and $k_{b2} = 1$. The control parameters are chosen as $\alpha = 0.6$ and $\beta = 2$.

The effectiveness of the designed algorithm is verified by two actuator failures.

Case 1: Incipient fault

The fault tolerance function is selected as follows:

$$\chi(x, u) = 5(x_1 x_2 + 0.3 \sin(u)) + 5 \quad (75)$$

The actuator failure time function is as follows:

$$\kappa(t - T_0) = \begin{cases} 0, & t < T_0 \\ 1 - e^{-8(t-T_0)}, & t \geq T_0 \end{cases} \quad (76)$$

The failure rate of unknown faults is $\nu = 8$ and the failure time is $T_0 = 10$.

To verify the superiority of the algorithm in this paper, the simulation results are compared with the literature [33], as shown in Figs. 1-4. The tracking error of the system is given in Figure 1, from which it can be seen that the system stabilizes within 0.43 seconds and still achieves accurate tracking when the actuator fails. However, the tracking errors reported in reference [33] are large, and the system is less fault-tolerant. Fig. 2 shows the change curve of controller u , and Fig. 3 shows the tracking curve of the desired trajectory x_d against x_1 . From the figure, it can be seen that at the 4th

second, the system fails, and the system state x_1 is always within the constraint. The curve for state x_2 is shown in Fig. 4, and it can be seen that x_2 is also always within the constraint interval. Finally, Fig. 5 shows the $\hat{\Theta}_1$ and $\hat{\Theta}_2$ curves. The simulation results show that the method in this paper has good tracking performance, which further validates the effectiveness of the method.

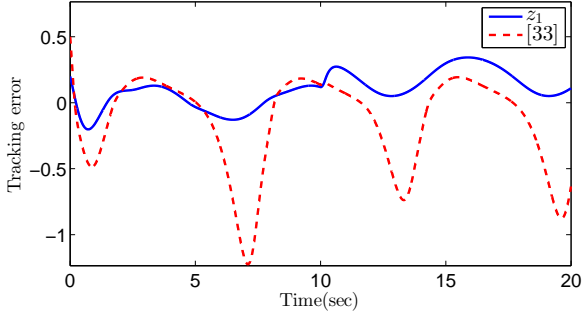


Fig. 1. Curve of the tracking error z_1

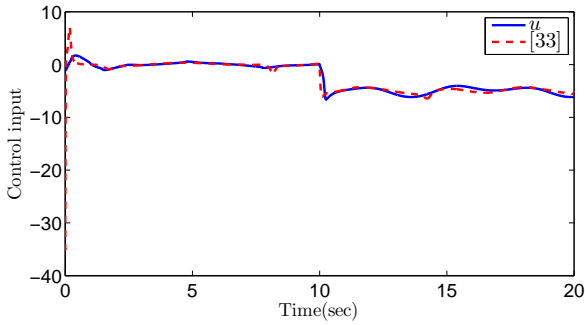


Fig. 2. Curve of the controller u

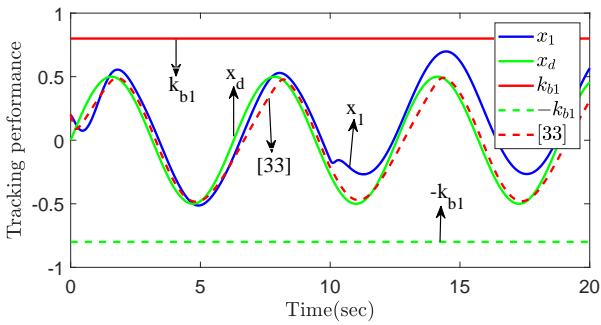


Fig. 3. Curve of state x_1 and desired trajectory x_d

Case 2: Abrupt fault

The system parameters are the same as those in the design for Case 1, except that the selection of the failure evolution rate is different. Here, the selection of ν is similar to a step function, which simulates sudden failure by choosing a larger value, i.e., $1 - e^{-\nu(t-T_0)}$ equals 1 if $\nu \rightarrow +\infty$.

Figs. 6-10 present the simulation results. From Fig. 6, it can be seen that the system has stabilized within 0.43 seconds, and accurate tracking can still be achieved when the system actuator fails. The simulation results show that the method has a good tracking effect in cases of sudden actuator failure. Meanwhile, Table 1 shows that the tracking

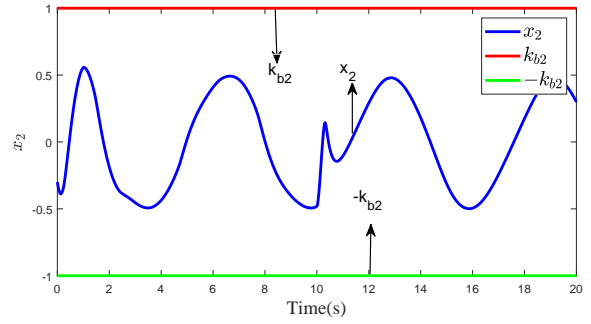


Fig. 4. Curve of x_2 and Constraint

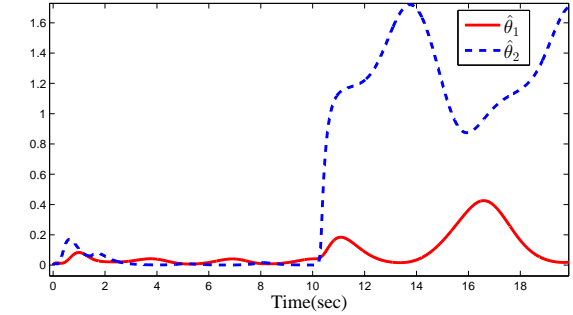


Fig. 5. Parameter estimation $\hat{\Theta}_1$ and $\hat{\Theta}_2$ curve

error convergence time of the algorithm in this paper is faster, and reference [33] does not consider the convergence time problem.

TABLE I
TRACKING PERFORMANCE COMPARISONS

| Method | Max tracking error | Error convergence time |
|--------|--------------------|------------------------|
| [33] | 1.25 | — |
| Ours | 0.3 | 0.43 |

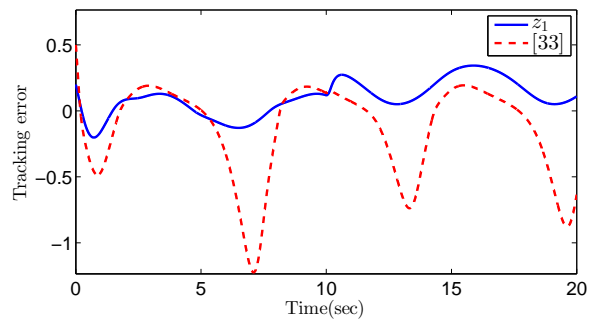


Fig. 6. Curve of the tracking error z_1

B. Physical example

We choose an actual electromechanical system, and its schematic diagram is shown in Fig. 11. The system model

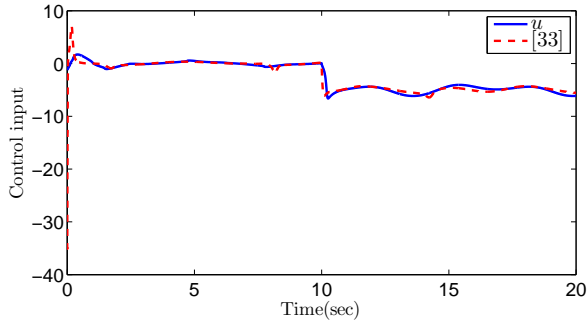


Fig. 7. Curve of the controller u

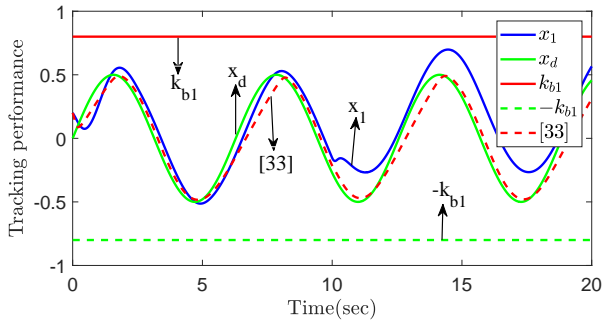


Fig. 8. Curve of state x_1 and desired trajectory x_d

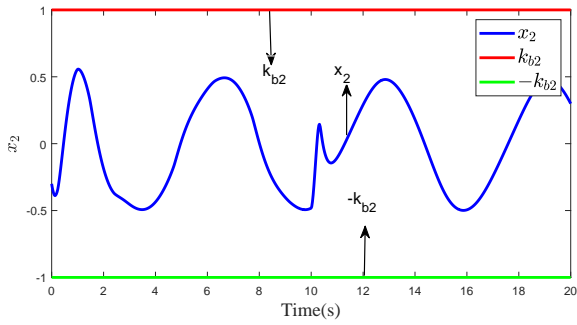


Fig. 9. Curve of x_2 and Constraint

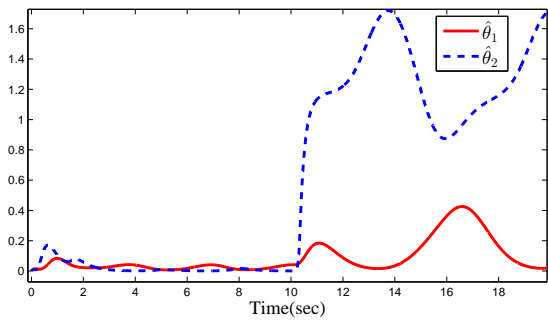


Fig. 10. Parameter estimation $\hat{\theta}_1$ and $\hat{\theta}_2$ curve

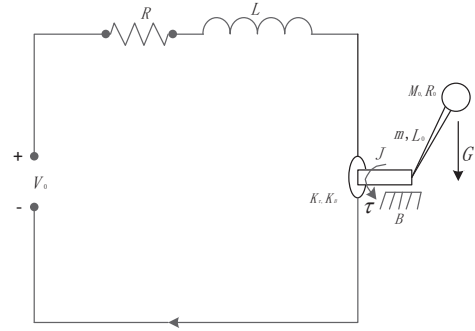


Fig. 11. Schematic of the electromechanical system

TABLE II
PARAMETERS OF THE ELECTROMECHANICAL SYSTEM AND THE UNKNOWN DISTURBANCE

| | |
|---|--|
| $J = 0.1625 \text{kg} \cdot \text{m}^2$ | $M_0 = 0.434 \text{kg}$ |
| $m = 0.506 \text{kg}$ | $R = 0.0005 \Omega$ |
| $G = 9.8 \text{N/kg}$ | $K_\tau = 0.9 \text{N} \cdot \text{m/A}$ |
| $K_B = 0.09 \text{N} \cdot \text{m/A}$ | $B_0 = 0.01625 \text{N} \cdot \text{m} \cdot \text{s/rad}$ |
| $L = 0.5 \text{H}$ | $d_1(x_1, t) = 0.5 \sin(x_1^2) + 0.01 \cos(0.1x_1 t)$ |
| $L_0 = 0.0305 \text{m}$ | $d_2(x_2, t) = 0.02 \cos(x_2 t)$ |
| $R_0 = 0.23 \text{m}$ | $d_3(x_3, t) = 0.03 \sin(x_3 t)$ |

expression can be expressed as follows

$$\begin{cases} \dot{x}_1 = x_2 + d_1(x_1, t) \\ \dot{x}_2 = \frac{1}{\frac{J}{K_\tau} + \frac{mL_0^2}{3K_\tau} + \frac{M_0L_0^2}{K_\tau} + \frac{2M_0L_0^2}{5K_\tau}} (x_3 - \frac{B_0}{K_\tau} x_2) \\ \quad - \frac{\frac{mL_0G}{2K_\tau} + \frac{M_0L_0G}{K_\tau}}{\frac{J}{K_\tau} + \frac{mL_0^2}{3K_\tau} + \frac{M_0L_0^2}{K_\tau} + \frac{2M_0L_0^2}{5K_\tau}} \sin(x_1) + d_2(x_2, t) \\ \dot{x}_3 = \frac{1}{L} u - \frac{K_B}{L} x_2 - \frac{R}{L} x_3 + d_3(x_3, t) \\ y = x_1 \end{cases} \quad (77)$$

where R is the armature resistance, $d_i(x_i, t)$ is the unknown disturbance, L is the armature inductance, J is the rotor inertia, K_B is the back-emf coefficient, m is the link mass, G is the gravity coefficient, V_0 is the input control voltage, M_0 is the load mass, L_0 is the link length, and R_0 is the radius of the load. B_0 is the coefficient of viscous friction at the joint and K_τ is the coefficient that characterizes the electromechanical conversion of armature current to torque. The above parameters of the electromechanical system and the unknown disturbance are shown in Table 2. Choose the desired trajectory as $x_d = \sin(t)$. The states are constrained by $k_{b1} = 1.2$, $k_{b2} = 5$, and $k_{b3} = 20$. The control parameters are chosen as $\alpha = 0.6$ and $\beta = 2$. Here, an abrupt actuator fault is considered, and the incipient fault treatment method is no longer elaborated on.

The simulation results are shown in Figs. 12-16, from which it can be seen that the states of x_1 , x_2 , and x_3 are still within the constraint intervals when an actuator sudden failure occurs in the electromechanical system. Meanwhile, the tracking error of the system is small and fault-tolerant.

V. CONCLUSION

The neural adaptive practical fixed-time control problem for strictly feedback nonlinear systems with full state constraints and actuator faults is investigated. Using the practical fixed-time theory and the backstepping method, it is demonstrated that the closed-loop system has desirable tracking

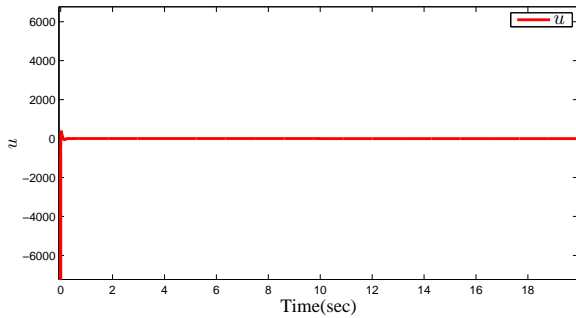


Fig. 12. Curve of the controller u

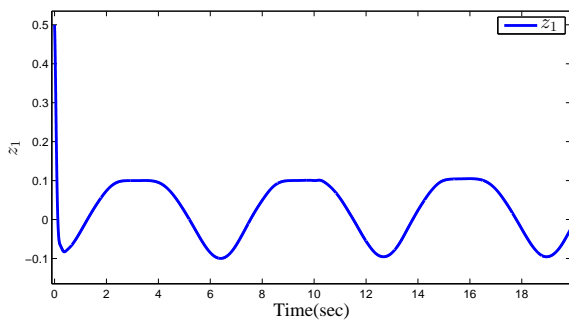


Fig. 13. Curve of the tracking error z_1

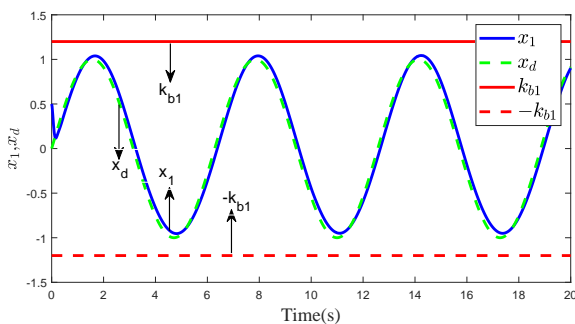


Fig. 14. Curve of state x_1 and desired trajectory x_d

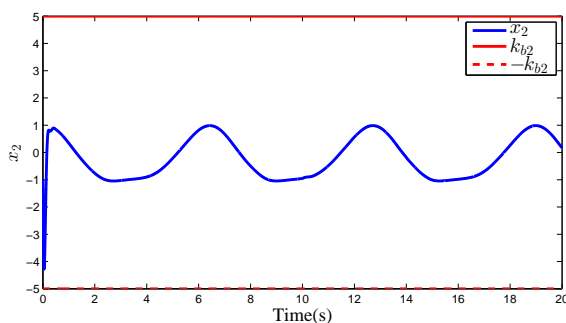


Fig. 15. Curve of x_2 and Constraint

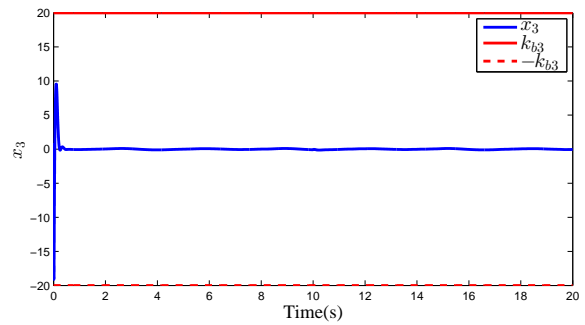


Fig. 16. Curve of x_3 and Constraint

performance under actuator faults and full-state constraints. Simulation results show that the method is fault-tolerant to actuator faults.

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